M.I.C.M. - C.N.I.A.R. INTREPRINDEREA DE CONSTRUCTII AERONAUTICE 2200 B R A S O V

MANDATORY SERVICE BULLETIN

IS-28B2/E0-7

Approved by : Department of Civil Aviation

with no. 345 / 8.01.1979

Product: IS-28B2 glider

Object : Additional indications concerning the

wing locking.

Compliance : Revision Record Card no.657.

DATE: 04.01.1979

IS-28B2/E0-5 Page 1/5

1. PLANNING INFORMATION

A. Applicability

This bulletin is applied by the users to gliders under servicing up to serial no.163 inclusive except the gliders serial no. 115, 119, 128, 140, 141, 143, 144, 145, 146, 147, 150, 151, 152, 153, 154, 155, 156, 157, 158, 160, 161 to which the modification was applied by the manufacturing plant.

To gliders beginning with serial no.164 the modification is applied by the manufacturing plants

This bulletin will be applied until the 1st of March 1979 and not later.

B. Reason

Additional indications concerning the wing locking.

C. Description

Applying the warning placard part no. 28B.F03.882/887 by gluing it on the wing spar in the junction area for exported gliders and part no. 28B.F03.883/888 for the gliders existing in Romania.

D. Compliance

Revision Record Card no.657.

E. Accomplishment

To gliders under manufacturing this bulletin is applied as current modification. To gliders under servicing it is applied by the users not later than the 1st of March 1979.

F. Material-Cost and Availability

Materials necessary to apply this bulletin are specified in chapter 3A and delivered by the manufacturing plant together with this service bulletin. Application belongs to the users.

DATE: 04.01.1979

- G. Tooling
- H. Weight and Balance
- I. References
 See Appendix I.
- J. Affected Documents

Replacing pages 02, 03A, 20, 2.11 and 6 from the Flight and Maintenance Manual 3rd issue April 1978 with pages 02, 03B, 20A, 2.11A, 6.0/3A and addition of page 6.4.1.A, issue 3.A, December 1978.

2. ACCOMPLISHMENT INSTRUCTIONS

- Dismounting the upper cowling between wings.
- B. Application

 Clean the application place of the warning placard with a soft cloth according to appendix I. Remove the backward protective skin and apply the label. Then firmly press the label with a seft cloth.
- C. <u>Using Instructions</u>
 None.

3. MATERIAL INFORMATION

- A. Material list

 warning placards
- B. Tooling list

DATE: 04.01.1979

C. Supply Indications

The placards are delivered by the manufacturing plant together with this bulletin.

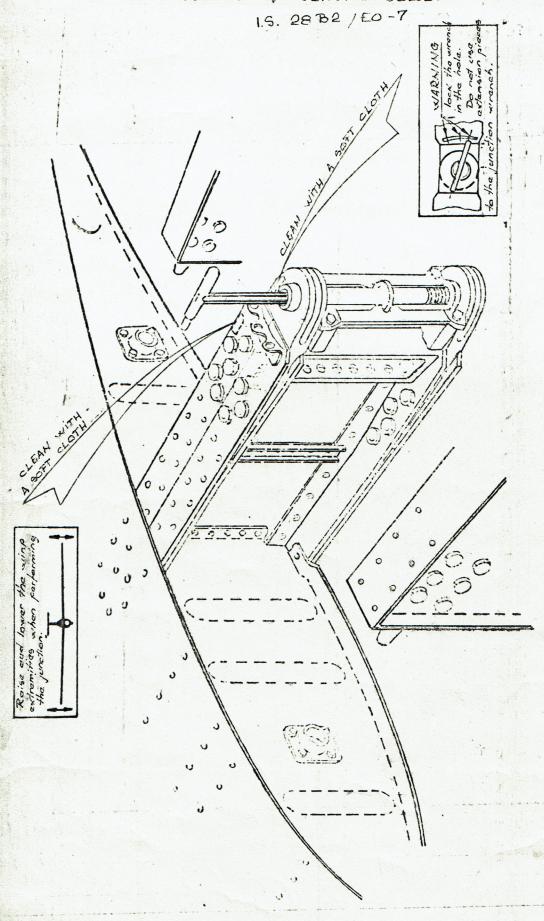
4. IDENTIFICATION

The modifications application is mentioned in the glider logbook.

5. APPENDICES

Warning placards according to chapter 3.A from the Mandatory Service Bulletin IS-28B2/E0-7.

Pages 02, 03B, 20A, 2.11.A, 6.0/3A and 6.4.1A issue 3A - December 1978.



REVISION RECORD CARD

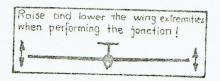
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Part no.	Amended page	Revision nature	Dat of appreval	a of insertion	Signature
1.	6.6.1.A 03.A	Tail junction bolts	04.12.78	08.12.78	
2.	03B; 2.0.A; 2.11.A; 6.0/3.A; 6.4.1.A	Additional indications concerning the junction lock	09.01.79	12.01.79	
3.	A.l.l A.l-A.l.l4	Dittel radio installation (optional)	11.05.79	14.05.79	
410	0.3.0; 2.1.A; 5.5.A; 6.14.A; 6.15.A; 6.46.A; 6.47.A; 6.48.A;	Additional indications	26.06.79	28.06.79	
5.	0.3.D; 2.3.A; 2.5.A; 2.7.A; 6.46.B;	Indications for light pilots	31.01,1980		
6.	0,3.E 1.0.A 1.5.A	Introducing of flush hinges for the glider serial no. 231 and subs.	12.02.80	19. 02 . 80	
7.	A.2.1; A.2.2.	Air brake on lower side (optional)			

3rd E ISSUE - NOVEMBER 1979

FLIGHT AND MAINTENANCE MANUAL IS-28B2

VALID PAGES LIST

2.9.4.9. Placard indicating the wing junction mounting (red coloured)



2

2.9.4.10. Placard indicating the wing junction locking (red coloured)



2.9.4.11. Other markings

- near the landing gear wheel:
- on landing gear oleo-pneumatic shock-absorber:
- near the tail skid wheel :
- on rudder, elevator and flaps
- leveling points and static pressure sources are marked.

25 atm (35 psi)

60-65 atm (840-910 psi)

25 atm (35 psi)

DO NOT PUSH

FLIGHT AND MAINTENANCE MANUAL IS-28B2

	Dra	awi	ngs:	Page
	Fig.	6-1	Fuselage-wing junction detail	6.3
	Fig.	6-2	Wing approach device	6.4
21	Fig.	6-2.1	Main junction tightening	6.4.1
	Fig.	6-3	Tails junction details	6.6
	Fig.	6-4	Manoeuvring wheel mounting	6.8
	Fig.	6-5	Greasing points	6.12
	Fig.	6-6	Releaser control	6.16
	Fig.	6-7	Landing gear	6.18
	Fig.	6-8	Landing gear retraction	6.19
	Fig.	6-9	Landing gear retraction control (for gli-	
			ders having a serial no.up to.31)	6.21
	Fig.	6-10	Landing gear shock-absorber	6.22
	Fig.	6-11	Inspection covers location	6.24
	Fig.	6-12	Elevator control	6.26
	Fig.	6-13	Aileron control	6.27
	Fig.	6-14	Rudder control	6.28
	Fig.	6-15	Trim tab control	6.30
	Fig.	6-16	Air brake and wheel brake control	6.31
	Fig.	6-17	Wheel brake control (for gliders with	
			serial no.up to 54)	6.33
	Fig.	6-18	Air brake control (for gliders with	
			serial no. up to 54)	6.34
	Fig.	6-19	Flap control	6.35
	Fig.	6-20	Flap control (for gliders with serial no.	
			up to 30)	6.36
	Fig.	6-21	Instruments connection diagram	6.37
	Fig.	6-22	Turn and Bank Indicator electrical diagram	6.39
	Fig.	6-23	Total energy variometer system	6.41

3rdA ISSUE : DECEMBER 1978

Page 6.0/3A

CHAPTER 2

LIMITATIONS

C o n	tent	3:		Page		
2.1.	Certification					
2.2.	Speed limitations					
2.3.	Airspeed indicator marking					
2.4.	Load factors and limit weights					
2.5.	Maximum weights					
2.6.	Balance					
2.7.	Accepted evolutions					
2.8.	Special instructions					
2.9.	Placards					
	2.9.1.	Limitatio	ons placard	2.4.		
	2.9.2. Aerobatic manoeuvres speed placard					
	2.9.3.	Warning P	placard for max.luggage into			
		the luggs	age compartment	2.8.		
	2.9.4.	Controls	identification placards	2.8.		
		2.9.4.1.	Placards for air brake and wheel brake control	2.8.		
		2.9.4.2.	Placards for flap control	2.9.		
		2.9.4.3.	Placards for trim tab control	2.9.		
		2.9.4.4.	Placards for landing gear retraction control	2.9.		
		2.9.4.5.	Placards for canopy opening and jettison	2.10.		
		2.9.4.6.	Placard for release control	2.10.		
		2.9.4.7.	Placard for the front air scoop control	2.10.		
		2.9.4.8.	Placard indicating the direction for wing main junction lock	2.10.		
	2	2.9.4.9.	Placard indicating the wing junction mounting	2.11A.		
		2.9.4.10	Placard indicating the wing junction locking	2.11A.		
		2.9.4.11	Other markings	2.11A.		

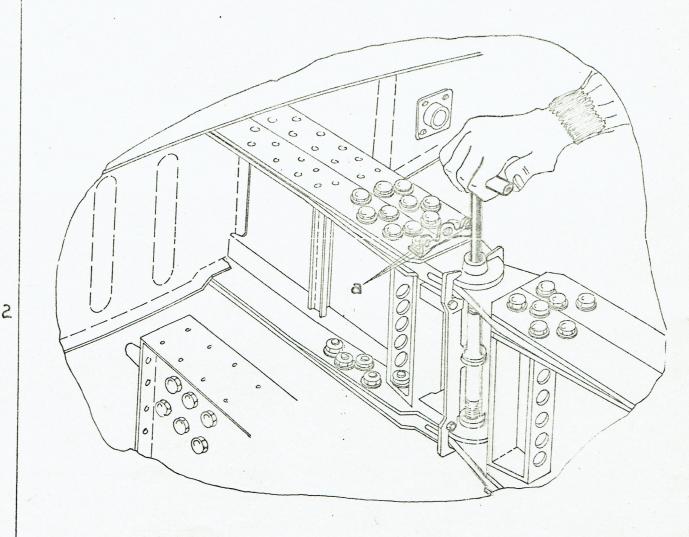


Fig. 6.2.1 Main junction tightening

The junction wrench shall be actuated by hand only (without using extensions).

During the whole tightening operation two persons shall raise and lower the wing extremities simultaneously to facilitate the cones lead in.

Match the wrench into one of the three locking holes indicated by "a" by removing, twisting and inserting it, until the correposition is found (sometimes a slight tightening is also required) The appended pages 02, 03B, 20A, 2.11.A, 6.0/3A and 6.4.1.A.issue 3A - December 1978 shall NOT BE INSERTED in the Flight
and Maintenance Manual 1st and 2nd issue.