

## MAJOR CHANGE APPROVAL

**10060677**

This Major Change Approval is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

### AEROCLUBUL ROMANIEI

LASCAR CATARGIU NR. 54  
010673 BUCURESTI  
ROMANIA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Type Certificate Number:** EASA.A.453

**Type Certificate Holder:** Aeroclubul Romaniei

**Type:** IS sailplanes

**Model:** IS-28B2

**Description of Design Change:**

IS-28B2 rear fuselage reinforcement in accordance with Aeroclubul Romaniei Master Document List 02.MDL.003, issue 05, dated 22. December 2016.

**EASA Certification Basis:**

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

**For the European Aviation Safety Agency**

**Date of Issue:** 10 January 2017



**Dominique ROLAND**  
Head of General Aviation and  
Remotely Piloted Aircraft Systems (RPAS)

10045624  
MAJOR CHANGE APPROVAL - 10060677 - AEROCLUBUL ROMANIEI - 307908

**Associated Technical Documentation:**

Aeroclubul Romaniei Master Document List 02.MDL.003, issue 05, dated 22. December 2016  
or later revisions of the above listed documents approved by EASA.

**Limitations/Conditions:**

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -